

Stepsize errors in the N -body problem: discerning Mercury's true possible long-term orbits

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ABSTRACT

Numerical integrations of the Solar System have been carried out for decades. Their results have been used, for example, to determine whether the Solar System is chaotic, whether Mercury's orbit is stable, or to help discern Earth's climate history. We argue that all of the past studies we consider in this work are affected by numerical chaos to different degrees, affecting the possible orbits and instability probability of Mercury, sometimes significantly. We show how to eliminate the effects of numerical chaos by resolving Mercury's pericentre passage. We also show that several higher order symplectic maps do not exhibit significant differences in resolving pericentre passage of Mercury (at fixed time step), making their advantages suspect for calculating long-term orbits. Resolving pericentre passage affects a wide array of orbital numerical studies, like exoplanet studies, studies of the galactic centre, and other N -body problems.

Key words: methods: numerical—celestial mechanics—planets and satellites: dynamical evolution and stability

1 INTRODUCTION

n -body problems (Heggie & Hut 2003) describe the motion of point particles interacting through forces that depend on pairwise distances. In astronomy, they are used to model planetary systems, galaxies, or star clusters, among other problems. For planetary systems described by conservative Hamiltonian systems, the Lie algebra structure is often used to derive numerical methods known as symplectic methods (Hairer et al. 2006). The most famous symplectic method is arguably the Wisdom–Holman (Wisdom & Holman 1991) method.¹ This mathematical structure is easily generalized to encompass non-conservative systems (e.g., Tamayo et al. 2020). Symplectic methods suffer from the well-known and studied roundoff errors and truncation errors. However, another source of error, arguably less understood, is that the stepsize chosen is limited by stepsize resonances with the physical frequencies in the system (Wisdom & Holman 1992). The onset of chaotic behavior can be qualitatively understood in terms of the overlap of resonances Chirikov (1979). The artificial resonances introduced by the symplectic map can overlap, introducing an additional source of unphysical chaos.

Wisdom & Holman (1992) showed how stepsize resonance overlap occurs in a 2 planet system. Resonances occur between the numerical time step and the difference of the planets' mean

motions. Rauch & Holman (1999) studied numerical chaos in the context of the Stark Problem. Wisdom (2015) further showed how resolving pericentre passage avoids effects of stepsize resonances for another two planet system and developed a simple criterion for the timestep required for pericentre resolution. Hernandez et al. (2020) studied statistics of 2-planet systems, finding accurate statistics when this criterion was respected.

Missing from these works are studies of numerical chaos in more realistic and complex problems like our Solar System, even if the extension is straightforward. Solar System integrations are used to determine the chronology of Earth's climate evolution over geologic timescales (e.g., Montenari 2018) and the evolution of orbits and chaos. The effects of resolving pericentre in such studies have not been analyzed, to our knowledge. Moreover, an analysis of numerical chaos in higher order and more sophisticated modern numerical methods beyond those used by Wisdom & Holman (1992); Rauch & Holman (1999); Wisdom (2015) is missing.

The aim of this work is to help fill this gap by studying the effects of pericentre resolution in Solar System integrations in the literature. Additionally, we carry out a study on a wider variety of numerical methods than the works above. We study three sets of Solar System data, finding they resolve pericentre of Mercury to different degrees. In particular, we find a data set that does not meet the criterion for pericentre resolution (Abbot et al. 2021) leads to an evolution of Mercury's eccentricity that is inconsistent at some times with another data set that resolves pericentre to a better degree (Zeebe 2015b). The paper is organized as follows. Section 2 gives an overview of the numerical methods used in this study and

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¹ The Wisdom–Holman map was not derived originally via the Lie algebra framework.

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carries out a study of how well they resolve pericentre. Section 3 describes Solar System integrations in the literature, and carries out first tests on pericentre resolution in the Solar System. Section 4 measures the effects of pericentre resolution on the evolution of Mercury's eccentricity for different data sets.

2 METHODS

Consider the n -planet problem, which describes perturbed Keplerian motion around a dominant mass. In this section we describe symplectic (Hairer et al. 2006; Channell & Scovel 1990; Hernandez 2019) and time-reversible (Hairer et al. 2006; Hernandez & Bertschinger 2018) maps, or integrators, typically used to study such systems in astrophysics over long time scales. We use notation described in more detail in Dehnen & Hernandez (2017). It also more fully gives mathematical background for our work. A version of the Wisdom–Holman method (Wisdom & Holman 1991; Kinoshita et al. 1991) can be written,

$$e^{h\hat{H}_{\text{WH}}} = e^{\frac{h}{2}\hat{A}} e^{h\hat{B}} e^{\frac{h}{2}\hat{A}}. \quad (1)$$

\hat{A} is an operator for function A . h is the fixed timestep. \hat{A} acts on functions of phase space f : $\hat{A}f = \{f, A\}$, where $\{\}$ indicate Poisson brackets. We have assumed, for simplicity, a conservative system with Hamiltonian $H = A + B$. The arguments in this paper are easily generalized to nonconservative systems described by first order differential equations, $\dot{x} = (\hat{A} + \hat{B})x$ (e.g., Tamayo et al. (2020); Laskar & Robutel (2001)). \hat{H}_{WH} is derived from the BCH formula (Hairer et al. 2006). A and B can be written in a variety of coordinate systems, and the functions A and B can be chosen differently, so long as $H = A + B$ (Hernandez & Dehnen 2017). A describes Keplerian motion and is integrable. B includes all other forces and effects in the system. Eq. (1) can, but need not, be modified further to produce efficient maps (e.g., Laskar & Robutel 2001; Hernandez & Dehnen 2017). Another modification is to separate the momentum dependent terms of B (if possible) as $B = B'(q) + S(p)$, where q and p are coordinates and momenta, respectively. Then, we can write:

$$e^{h\hat{H}_{\text{WH}}} = e^{\frac{h}{2}\hat{A}} e^{\frac{h}{2}\hat{S}} e^{h\hat{B}'} e^{\frac{h}{2}\hat{S}} e^{\frac{h}{2}\hat{A}}. \quad (2)$$

Let ϵ be the ratio of the maximum planetary mass to the dominant mass. The error of WH is then $O(\epsilon h^2)$.

We denote the next method by WHc, a “corrected” (Wisdom et al. 1996; Wisdom 2006) version of WH.

$$e^{h\hat{H}_{\text{WHc}}} = e^{\hat{C}} e^{\frac{h}{2}\hat{A}} e^{h\hat{B}} e^{\frac{h}{2}\hat{A}} e^{-\hat{C}}, \quad (3)$$

\hat{C} and $-\hat{C}$ need only be applied at input and output, respectively; this practically incurs no computational penalty. \hat{C} is constructed to an order m , and the error is $O(\epsilon h^{m+1} + \epsilon^2 h^2)$. For m large, the $O(\epsilon h^{m+1})$ error can be ignored. If the function $\delta = 1/24\{B, \{B, A\}\}$ is integrable, WHc can be improved further, to WHm. In particular, δ is integrable in the common case where A is quadratic in canonical momenta and B depends only on coordinates. WHm has higher compute cost than WHc and is written,

$$e^{h\hat{H}_{\text{WHm}}} = e^{\hat{C}} e^{\frac{h}{2}\hat{A}} e^{hB+h^3\delta} e^{\frac{h}{2}\hat{A}} e^{-\hat{C}}. \quad (4)$$

$h^2\delta < B$ for reasonable timesteps not affected by stepsize chaos. The error of WHm is $O(\epsilon h^{m+1} + \epsilon^2 h^4)$.

Next, we introduce Y, which combines ideas from Yoshida (1990) with a WH integrator:

$$e^{h\hat{H}_Y} = e^{\alpha h\hat{H}_{\text{WH}}} e^{\beta h\hat{H}_{\text{WH}}} e^{\alpha h\hat{H}_{\text{WH}}}. \quad (5)$$

Here, $\alpha = 1/(2 - 2^{1/3})$ and $\beta = -2^{1/3}/(2 - 2^{1/3})$. The error of Y is $O(\epsilon h^4)$, as can be checked numerically. The integrator package HNBODY implements a method equivalent to eqs. (2) and (5) in eqs. (8) and (5) in the documentation³, if certain forces are ignored.

Y can also be corrected, let this method be Yc. Yc is written,

$$e^{h\hat{H}_{Yc}} = e^{\hat{C}_Y} e^{\alpha h\hat{H}_{\text{WH}}} e^{\beta h\hat{H}_{\text{WH}}} e^{\alpha h\hat{H}_{\text{WH}}} e^{-\hat{C}_Y}. \quad (6)$$

The form of \hat{C}_Y has not been published. The error of Yc is $O(\epsilon h^{m+1} + \epsilon^2 h^4)$ (Kevin Rauch, private communication). For m large, Yc has the same error scaling as WHm. Note WH, WHc, and WHm (WH methods) share the same operator building blocks, so we might expect similarities in the phase space of orbits computed with them. Y and Yc (Y methods) also share similar building blocks. When we iterate WH maps, such as (1) or (2), over two or more timesteps, note the coefficient in front of \hat{A} is h (ignoring the first and last operators in the first and last steps, respectively). For Y methods, the coefficient in front of \hat{A} is $h/2$, so the Kepler problems are solved for slightly shorter times, and are thus better resolved.

Finally, consider LR, referred to as SABA4 by Laskar & Robutel (2001):

$$e^{h\hat{H}_{\text{LR}}} = e^{c_1 h\hat{A}} e^{d_1 h\hat{B}} e^{c_2 h\hat{A}} e^{d_1 h\hat{B}} e^{c_1 h\hat{A}}. \quad (7)$$

$c_2 \approx 0.577$, $d_1 = 1/2$, and $c_1 \approx 0.211$. The error is $O(\epsilon h^8 + \epsilon^2 h^2)$. The largest coefficient in front of any operator, when we iterate LR, is 0.577, suggesting it has a smaller effective timestep than WH. Indeed, in experiments with LR, Hernandez et al. (2020) found LR can use larger timesteps than WH and still give correct orbital phase space statistics.

We present a solution proposed by Laskar & Robutel (2001) to use LR with B nonintegrable. First, assume B can be decomposed into integrable pieces, $B = \sum_{i=1}^N B_i$. For k a constant, use the first order accurate substitution, $e^{k\hat{B}_1} e^{k\hat{B}_2} \dots e^{k\hat{B}_N}$, or $e^{k\hat{B}_N} e^{k\hat{B}_{N-1}} \dots e^{k\hat{B}_1}$ for $e^{k\hat{B}}$ such that map (7) stays symmetric. These substitutions allow the LR error to remain as $O(\epsilon h^8 + \epsilon^2 h^2)$. Next, we describe a corrector that reduces the error of LR, at additional expense. If one of the B_i ⁴ depends only on canonical positions, while A is quadratic in the canonical momenta, let $\delta_{\text{LR}} = 1/(2c)\{A, B_i\}, B_i\}$, where $c \approx 0.003397$ (Laskar & Robutel 2001). Then the method,

$$e^{h\hat{H}_{\text{LRm}}} = e^{-h\delta_{\text{LR}}} e^{h\hat{H}_{\text{LR}}} e^{h\delta_{\text{LR}}}, \quad (8)$$

eliminates the $\{A, B_i\}, B_i\}$ error terms, which scale as $\epsilon^2 h^2$. The various methods described here have different merits as far as efficiency goes; i.e., compute cost vs errors committed. This paper is not preoccupied with the efficiencies; various other works have investigated this (e.g., Rein et al. 2019; Wisdom 2018).

2.1 Pericentre resolution of methods

Nonlinear stability due to pericentre resolution has been investigated for simple problems and methods (Rauch & Holman 1999; Wisdom 2015). Here, we extend this analysis for different WH and Y methods, and investigate pericentre resolution over time.

³ <https://www.astro.umd.edu/~hamilton/research/new/RauHam.pdf>

⁴ It is easy to generalize this argument if more than one B_i satisfy this condition.

² Hernandez & Dehnen (2017) use a different convention for ϵ .

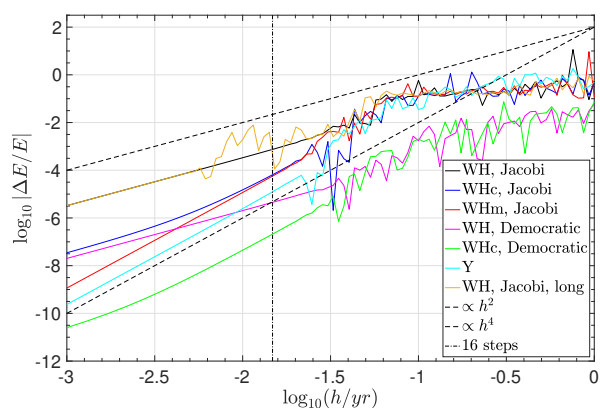


Figure 1. Nonlinear stability of WH and Y integrators. The two-planet problem is integrated for 1000 yrs for different stepsizes. The integrations described by the yellow curve are performed for 10000 yrs. Pericentre resolution for the 1000 yr integrations is well-described by the vertical line, representing the $\tau_f/16$ criterion.

Wisdom (2015) proposed that when the stepsize is larger than about $1/16$ of the effective period at pericentre (EPP), pericentre is no longer resolved for several problems. The EPP is defined as,

$$\tau_f = 2\pi \sqrt{\frac{(1-e)^3 a^3}{1+e GM}}, \quad (9)$$

where e is the eccentricity, a is the semi-major axis, M is the self-gravitating mass, and G is the gravitational constant.

We consider a first numerical test, the two-planet problem described in Levison & Duncan (2000); Wisdom (2015). This problem consists of Sun, Jupiter, and Saturn, with masses (1, 1/1047.355, 1/3498.5) Solar masses. The inclinations of Jupiter and Saturn are 0 and $\pi/2$, respectively. The arguments of perihelion and longitudes of ascending node are set to 0. The semi-major axes of Jupiter and Saturn are 5.2 au and 9.58 au, respectively. We use $G = 39.4845$ in units of Solar mass, yr, and au. This problem is integrated for 1000 years with various methods for different timesteps. The implementations of the numerical methods are our own. The final energy error as a function of timestep is shown in Fig. 1. We show different methods, both in Jacobi and Democratic Heliocentric coordinates (for a review, see Hernandez & Dehnen (2017)). The splitting for Democratic Heliocentric Coordinates and Jacobi coordinates is labelled WHD and WHJ, respectively, in Hernandez & Dehnen (2017). The “original form” gravitating mass for WHJ, as described in Hernandez & Dehnen (2017), is used. Y is composed of WHJ maps. The orange curve shows an integration 10 times longer, for 10000 yrs. For small timesteps, the error of all integrations is dominated by truncation error, and scales as h^2 or h^4 . When the error is no longer described well by truncation error, pericentre is no longer resolved. The criterion $\tau_f/16$ is indicated by a vertical line. For the 1000 yr integrations, this line is a conservative estimate of the transition to the stepsize resonance region. All methods transition at approximately the same timestep. This agrees with Hernandez et al. (2020), who compared the probability distribution functions (PDFs) of test particle orbits in the restricted three-body problem. They found the PDFs broke down at approximately the same time step for a variety of WH methods, where pericentre is no longer resolved. There are minor differences between the methods, which will be explored in more detail in Section 3.

We see the pericentre resolution requires smaller timesteps for the long integration. There is no violation of the $\tau_f/16$ rule; what happens in this case is that, for small timesteps, Saturn’s eccentricity has significantly increased from 0.950 at the end of the short simulations, to 0.979 at the end of the long simulations. This causes $\tau_f/16 = h$ at $\log_{10}(h/\text{yr}) = -2.4$, which is, to a good approximation, where we see pericentre resolution break down in Fig. 1 for the long integration.

3 SOLAR SYSTEM INTEGRATIONS IN THE LITERATURE

While the simple numerical experiments of Section 2.1 are helpful for getting a grasp on criteria for pericentre resolution, we wish to see the effects of stepsize chaos and diffusion on Solar System integrations, which has not been investigated previously to our knowledge. We consider three sets of Solar System integrations by different authors which may have different degrees of pericentre resolution of Mercury.

The first set of integrations were carried out by Zeebe (2015b) (Z15). The Solar System model includes the 8 planets, Pluto, point mass Newtonian gravity, and general relativistic correctors as described in Saha & Tremaine (1994). The method is Yc, with WH version (2), as implemented in HNBODY. Jacobi coordinates are used, and $m = 6$ for the correctors. 1600 Solar System integrations with slightly different initial conditions are performed. Zeebe (2015b) changes the timestep as Mercury’s eccentricity becomes large. Specifically, for low eccentricities $h = 4$ days is used. For $0.55 < e < 0.70$, $h = 1$ day is used by default, but see below. For $0.70 < e < 0.80$, $h = 1/4$ days is used. For higher eccentricities, an adaptive stepping (non symplectic) method is used.

For eccentricities (0.2, 0.55, 0.70, 0.80), the corresponding $\tau_f/16$ are (3.59, 1.33, 0.643, 0.35) days, respectively. We have assumed Mercury’s semi-major axis remains fixed at 0.387 au. To check this assumption, see Zeebe (2015a), Fig. 1c, where even for high eccentricities nearing $e = 0.8$, Mercury’s semi-major axis varies by approximately one part in 10^4 . It appears the timesteps do not satisfy the $\tau_f/16$ criterion for $e < 0.7$. However, the criterion is not violated as strongly as we have suggested here: on most of the integrations with high-eccentricity runs, Zeebe (2015b) actually reduces the timestep from $h = 4$ days to $h = 1$ day at eccentricities between ~ 0.4 and ~ 0.5 , rather than 0.55. Also, recall the coefficient in front of the \hat{A} operator is half of that of WH methods, which may modify the criterion. Integrations where the eccentricity of Mercury becomes larger than 0.55 are rare; only 1.75% of Z15 integrations have $e > 0.55$.

Next, we consider the integrations by Brown & Rein (2020). 96 integrations of the Solar System are performed with WHm and $m = 16$ for the corrector. Their software implementation is called WHCKL (Rein et al. 2019). They use Jacobi coordinates and $h \approx 8.062$ days. Their model has the 8 planets and the general relativistic model of Nobili & Roxburgh (1986), which introduces an error in the mean motion (Saha & Tremaine 1992). The extra Hamiltonian terms are only position dependent allowing use of WH version (1). They state their integrations, “can be trusted as long as the eccentricity remains moderate, $e \lesssim 0.4$.” Their stepsize is more than twice that of Z15’s, and the difference is even higher for higher Mercury eccentricities. Thus we expect Brown & Rein (2020) integrations to have worse pericentre resolution. Abbot et al. (2021) (A+21) perform 1008 Solar System integrations using the same integration configuration and software package as Brown & Rein

(2020). Thus, we can consider A+21 as an extension to the simulations of [Brown & Rein \(2020\)](#).

The final set of integrations we consider are those by [Laskar & Gastineau \(2009\)](#) (LG09). LG09 uses a Solar System model of the 8 planets, Pluto, general relativity, as suggested by [Saha & Tremaine \(1994\)](#), and several other small forces and effects such as the Earth-moon tidal dissipation and the Solar quadrupole moment. This is the only nonconservative set of integrations. They carry out 2501 Solar System integrations using LRm and $h \approx 9.13$ days. They reduce their timestep for $e > 0.4$, but appear not to state to what value. Their B is non-integrable, but can be decomposed into integrable parts as discussed in Section 2. Out of the three Solar System data sets we have considered so far, LG09 use the largest timestep. However, recall from Section 2 that the effective stepsize for LRm can be considered $\approx 0.577h$, and we conclude that LG09 uses the second largest effective time step for small eccentricities.

Given their stepsizes, we assume the datasets, ranked from most to least degree of pericentre resolution are Z15, LG09, and A+21. We can find some supporting evidence for this already from [Abbot et al. \(2021\)](#), Figure 3. They compare the Z15, LG09, and A+21 integrations. [Abbot et al. \(2021\)](#) compute a Mercury instability probability, finding the datasets, from most to least unstable, are A+21, LG09, and Z15. Because a failure to resolve pericentre leads to more instability and possibly diffusion, this agrees with our prediction. They find the instability probability of LG09 and Z15 agree within error bars. For $t \gtrsim 3.5$ Gyrs, the A+21 instability probabilities are not consistent with the LG09 and Z15 probabilities. We will revisit this in Section 4.

3.1 Analysis of Z15 integrations

To see the effects of resolving Mercury's pericentre on Solar System integrations, we consider the model and integrators used by Z15, described in Section 3. The initial conditions are taken from the 299th run, when Mercury's eccentricity has reached $e = 0.84$, so as to minimize the EPP. The initial conditions are integrated 1000 yrs using Yc with WH version eq. (2), and $m = 6$. Different timesteps are used. WHc is also used with $m = 6$. The implementations of HNBody are used. Mercury's semi-major axis varies by $|\Delta a/a| \leq 10^{-4}$ in the integrations.

Root-mean-square energy errors as a function of timestep are plotted in Fig. 2. Also indicated is the $\tau_f/16$ criterion, and slopes $\sim h^2$ and h^4 . It appears errors from stepsize resonance present itself at smaller timesteps for Yc, in agreement with comparisons between low and high order methods in [Wisdom \(2015\)](#). In reality, both methods exhibit errors from pericentre passage at approximately the same step, but the WHc errors are too small to be seen on the scale of Fig. 2 for small timesteps.

To see this, in Fig. 3, we plot instead the average errors as a function of timestep. Now, the average errors are dominated by roundoff error for small timesteps, and transition to a region dominated by stepsize resonances at approximately the same time step, in agreement with the $\tau_f/16$ criterion. We have also run this problem without Mercury and verified that the effects of stepsize resonances disappear for the timestep ranges and the scale in error of Fig. 3.

In our experiments, the evolution of stepsize chaos with time is small and insignificant. This is consistent with [Hernandez et al. \(2020\)](#), Fig. 10. They studied the variance of the Jacobi constant error over time for restricted three-body problems. This variance grew linearly for symplectic integrators with moderate timesteps.

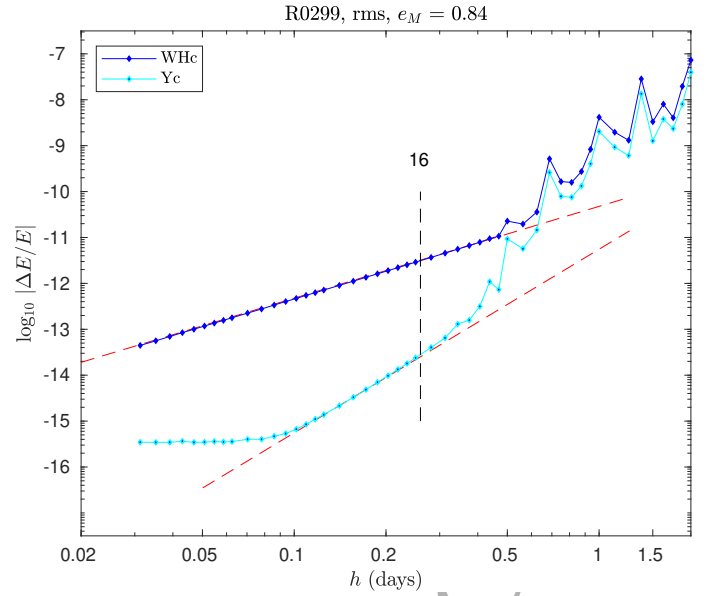


Figure 2. Integration for 1000 yrs and various timesteps of a Solar System with a highly eccentric Mercury. Yc and WHc are used, and root-mean-square errors as a function of time are plotted. A vertical line indicates $\tau_f/16$. Truncation error dominates over errors from stepsize resonance at small timesteps for WHc.

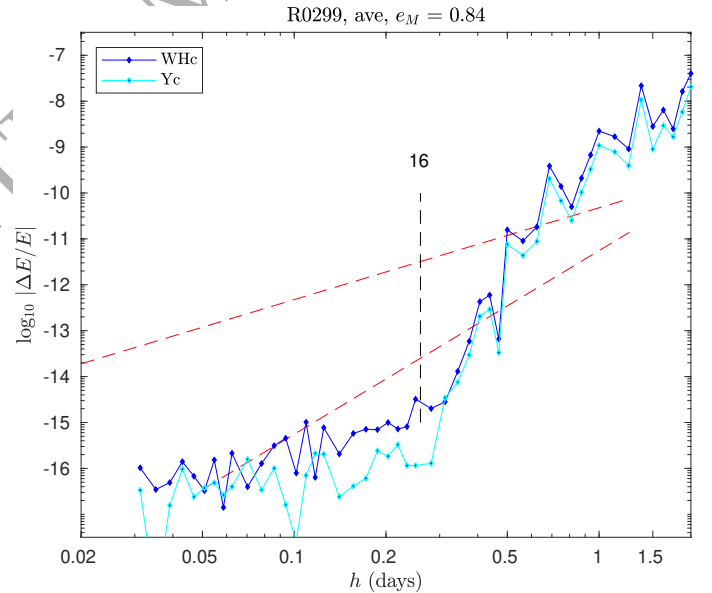


Figure 3. Same as Fig. 2, but average errors are plotted instead. WHc and Yc are now seen to transition to the region dominated by roundoff error at similar timesteps predicted by $\tau_f/16$.

For small timesteps, there was no clear trend in the evolution of the variance.

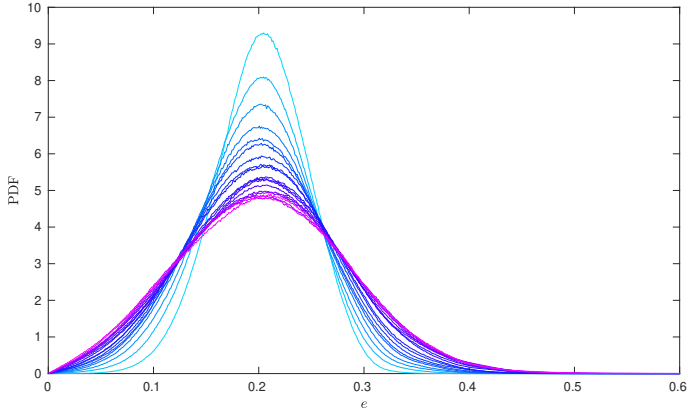


Figure 4. PDFs of Mercury's eccentricity over time from the Z15 data. A PDF is plotted every 250 Myrs. The first 250 Myr time slice is not shown. The colors towards the purple end of the spectrum indicate longer integration times and show Mercury's eccentricity diffusing.

4 INFLUENCE OF PERICENTRE RESOLUTION ON MERCURY'S ECCENTRICITY

The diffusion of Mercury's eccentricity is one mechanism for encountering Solar System instabilities (Laskar 2008). Here, we compare ensembles of N -body integrations with different time steps to explore how resolution of Mercury's EPP influences this chaotic diffusion; specifically, we will compare A+21 and Z15. In Fig. 4, we compute the probability density function (PDF) of Mercury's eccentricity every 250 Myrs, using the Z15 data sets. The PDF data for the first 250 Myrs are discarded. The colors towards the purple end of the spectrum indicate longer integration times and show Mercury's eccentricity diffusing (Batygin et al. 2015). A sampling rate of ~ 25 kyrs is used and the eccentricities are sorted into 700 bins between eccentricities 0 to 0.7. This figure can be compared to Laskar (2008), Figure 6.

To quantify the time evolution of Mercury's eccentricity in the two ensembles of integrations, we fit a two-parameter Rice distribution, with a probability density given by

$$f_{\eta,m}(x) = \frac{x}{\eta^2} \exp\left(-\frac{x^2 + m^2}{2\eta^2}\right) I_0\left(\frac{xm}{\eta^2}\right), \quad (10)$$

where I_0 is the modified Bessel function of the first kind, to the distribution of Mercury's eccentricity at different times in each ensemble. Note we are reusing the variable m , so that our notation is consistent with Laskar (2008).

We derive maximum-likelihood estimates for the distribution parameters, m and η^2 , by maximizing the likelihood function $\mathcal{L}(m, \eta^2 | \{e_i\}_{i=1}^N) \propto \prod_{i=1}^N f_{\eta,m}(e_i)$ for the set of N values of Mercury's eccentricity, $\{e_i\}_{i=1}^N$, at a given time, T , among the simulations in an ensemble. Note the samples obtained in this way are composed of approximately 1008 and 1600 eccentricity values per time slice for A+21 and Z15, respectively (simulations that experienced instabilities prior to the time T are removed from the ensembles before fitting for distribution parameters). A Z15 data point at ≈ 2.6 Gyrs is missing as output was not saved at that time.

We also obtain estimate uncertainties in the inferred parameters by computing the Hessian of the log-likelihood function at the maximum likelihood value of the parameters and using the Laplace approximation. In other words, we assume our maximum likelihood estimates of the parameters m and η^2 are approximately

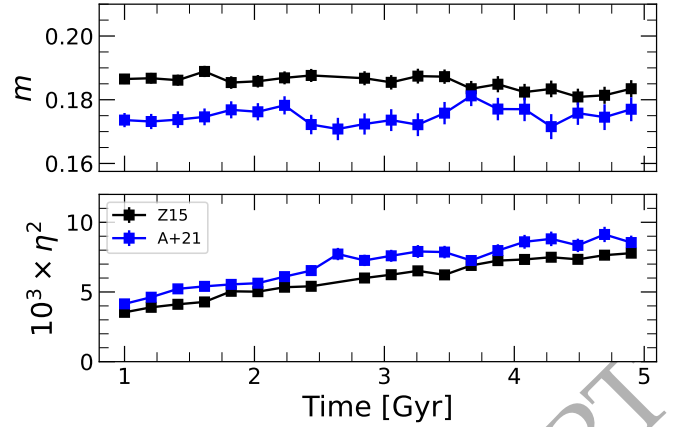


Figure 5. Comparison of the Z15 (black) and A+21 (blue) PDFs of Mercury's eccentricity as a function of time. Each panel shows a parameter of the Rice distribution, Equation (10), fitted to the sample of Mercury's eccentricity values taken from each ensemble of simulations at the plotted times. Error bars indicate estimated 1σ uncertainties in parameter values.

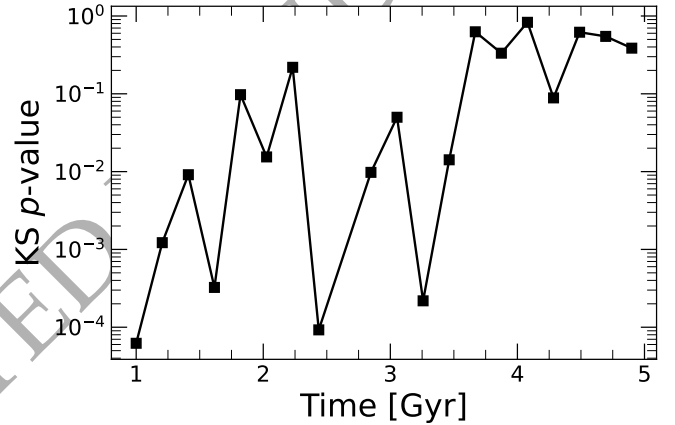


Figure 6. p -values from a two-sample KS test comparing the A+21 and Z15 simulation PDFs.

Gaussian-distributed about the true values with covariance matrix $\Sigma = -[\nabla_{(m,\eta^2)}^2 \ln \mathcal{L}]^{-1}$.

The parameter η is a proxy for the Rician width. η^2 is statistically significantly larger for A+21, as a function of time. This may be consistent with A+21, who find in their Figure 3 that A+21 has higher instability probabilities than Z15 as a function of time. They demonstrate their instability probabilities are correlated with eccentricity range. η and eccentricity range are likely correlated as well. m , which indicates the centering of the PDFs, is significantly smaller for A+21 for almost all time.

A two-sample KS test between the A+21 and Z15 data provides a more direct test of the compatibility of Mercury's eccentricity distribution in the two simulation ensembles without imposing any assumptions about an underlying functional form. In Fig. 6, we plot p -values from two-sample KS tests as a function of time. The p -values are small (< 0.05) for most data at time $t < 3.5$ Gyrs, indicating inconsistent simulations. However, for time $t > 3.5$ Gyrs, the p -values are large (> 0.05), indicating consistent PDFs between

A+21 and Z15. One might wonder how sensitive our KS p -values are to our choice of time slices. To explore this, we have recalculated Fig. 6 twice, first shifting the times by -100 Myr and then by $+1$ Myr. There are variations in the p -values, but the general trends remain: at times > 3.5 Gyr, the PDFs are mostly consistent, while at earlier times they are mostly inconsistent. This is the opposite trend observed by A+21 in their Figure 3, who find the instability probabilities of A+21 and Z15 become inconsistent for $t \gtrsim 3.5$ Gyrs.

5 DISCUSSION

We studied the effects of numerical chaos on Solar System integrations in the literature. We find significant differences in the evolution of Mercury's eccentricity between the data sets of Zeebe (2015b) and Abbot et al. (2021), and we postulate this is due to different levels of pericentre resolution. Our results are also supported by the study of Abbot et al. (2021), which compares Solar System instability probabilities between different data sets. We conclude that a failure to resolve pericentre can lead to unphysical orbits in dynamical systems in astronomy. However, we believe the comparison between the A+21 and Z15 data sets can be strengthened by increasing the number of integrations in each data set. Stepsize resonance errors can be avoided by adequately resolving the effective period at pericentre, Eq. (9). For a wide range of numerical methods (except a Laskar & Robutel (2001) method), the tests here and in other works (Wisdom 2015; Hernandez et al. 2020) suggest a time step of $\tau_f/16$ adequately avoids stepsize chaos, although this criterion may need slight modification for some problems (Wisdom 2015).

We note none of the Solar System data sets we have considered here strictly resolve pericentre; some are closer to resolving it than others, which explains the different instability probabilities of Abbot et al. (2021). It may be that Mercury's eccentricity evolution has not converged yet and looks different for an integration that uses smaller stepsizes than any considered here. Such stepsizes might better avoid numerical chaos and diffusion. Also note that the energy of a Keplerian orbit is independent of the eccentricity, so energy errors do not directly tell us about the eccentricity evolution of Mercury.

Lower order symplectic integrators can resolve Mercury's pericentre at lower computational cost compared to several higher order methods. This is a disadvantage of using higher order methods instead of lower order methods for Solar System calculations, and raises more questions about the usefulness of higher order symplectic integrators for planetary dynamics (Hernandez et al. 2021).

Finally, we mention that pericentre resolution is not a problem unique to planetary systems. Indeed, we expect resolving pericentre to be important in a variety of N -body systems, such as stellar clusters and galaxies. Future work can explore this issue.

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7 DATA AVAILABILITY

The data underlying this article will be shared on reasonable request to the corresponding author.

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